# 4.1 Design of Members

This section covers the following topics

- Calculation of Demand
- Design of Sections for Axial Tension

### **Introduction**

The design of prestressed concrete members can be done by the limit states method as given in **Section 4** of **IS:1343 - 1980.** 

First, the force demand in a member under the design loads is determined from a structural analysis. A preliminary size of the member is assumed for analysis. Next, the member is designed to meet the demand. If necessary, another cycle of analysis and design is performed.

The following material explains the calculation of the demand in a member under the design loads.

# **4.1.1 Calculation of Demand**

In the limit states method, the design loads are calculated from the characteristics loads by multiplying them with load factors (<sup>γ</sup>*f*). Several types of loads are considered to act together under the selected load combinations. The load factors are included in the load combinations as weightage factors.

The demand in a member for a particular type of load is obtained from the analysis of the structure subjected to the **characteristic value** of the load. The demands for the several load types are then combined under the load combinations, based on the principle of superposition.

### **Characteristics Loads**

For dead loads, a characteristic load is defined as the value which has a 95% probability of not being exceeded during the life of the structure. This concept assumes a normal distribution of the values of a particular dead load. In the following figure, the shaded area above the characteristic value represents 5% probability of exceedance of the load in the design life of the structure.



**Figure 4-1.1** Idealised normal distribution for a dead load

For live load, wind load and earthquake load, a characteristic load is defined based on an extreme value distribution. For example, the characteristic wind load is defined as the value which has a 98% probability of not being exceeded during a year.



The characteristics loads can be obtained from **IS:875 - 1987** (*Code of Practice for Design Loads for Buildings and Structures*) and **IS:1893 - 2002** (*Criteria for Earthquake Resistant Design of Structures*) as follows.





For special loads, there are some guidelines in **IS: 875 - 1987, Part 5**. In addition, specialised literature may be referred to for these loads. The special loads are listed below.

**Temperature Hydrostatic** Soil pressure **Fatigue** Accidental load Impact and collision **Explosions Fire** 

For special situations, the loads are determined from testing of prototype specimens. Dynamic load tests, wind tunnel tests, shake table tests are some types of tests to determine the loads on a structure. Finite element analysis is used to determine the stresses due to concentrated forces and dynamic loads.

### **Load Factors and Load Combinations**

The load factors and the combinations of the various types of loads are given in **Table 5** of **IS:1343 - 1980**. The following are the combinations for the ultimate condition.

 $1.5$  (DL + LL) 1.2 ( $DL + LL \pm WL$ ) 1.2 ( $DL + LL \pm EL$ ) 1.5 ( $DL \pm EL$ )

1.5 ( $DL \pm WL$ )  $0.9$  DL  $\pm$  1.5 EL

The load combinations for service conditions are as follows.

DL + LL  $DL + 0.8$  (LL  $\pm$  EL)  $DL \pm EL$  $DL \pm WL$ 

#### **Analysis of Structures**

Regarding analysis of structures, **IS:1343 - 1980** recommends the same procedure as stated in **IS:456 - 2000**. A structure can be analysed by the linear elastic theory to calculate the internal forces in a member subjected to a particular type of load.

#### **Design of Members**

There can be more than one way to design a member. In design, the number of unknown quantities is larger than the number of available equations. Hence, some quantities need to be assumed at the beginning. These quantities are subsequently checked.

The member can be designed either for the service loads or, for the ultimate loads. The procedure given here is one of the possible procedures. The design is based on satisfying the allowable stresses under service loads and at transfer. Initially, a lumpsum estimate of the losses is considered under service loads. After the first round of design, detailed computations are done to check the conditions of allowable stresses. Precise values of the losses are computed at this stage. The section is then analysed for the ultimate capacity. The capacity should be greater than the demand under ultimate loads to satisfy the limit state of collapse.

### **4.1.2 Design of Sections for Axial Tension**

#### **Introduction**

Prestressed members under axial loads only, are uncommon. Members such as hangers and ties are subjected to axial tension. Members such as piles may have bending moment along with axial compression or tension.

#### **Design of Prestressing Force**

First, a preliminary dimension of the member is selected based on the architectural requirement. The prestressing force at transfer  $(P_0)$  should be such that the compressive stress in concrete is limited to the allowable value. At service, the designed prestressing force (*Pe*) should be such that the tensile stress in concrete should be within the allowable value. The amount of prestressing steel  $(A<sub>o</sub>)$  is determined from the designed prestressing force based on the allowable stress in steel.

**At transfer**, in absence of non-prestressed reinforcement, the stress in concrete (*fc*) is given as follows.

$$
f_c = -\frac{P_0}{A_c} \tag{4-1.1}
$$

Here,

 $A_c$  = net area of concrete

 $P_0$  = prestress at transfer after short-term losses.

In presence of non-prestressed reinforcement, the stress in the concrete  $(f_c)$  can be calculated as follows.

$$
f_c = -\frac{P_0}{A_c + (E_s/E_c)A_s}
$$
 (4-1.2)

Here,

*As* = area of non-prestressed reinforcement

*Es* = modulus of elasticity of steel

*Ec* = modulus of elasticity of concrete.

At service, the stress in concrete ( $f_c$ ) can be calculated as follows.

$$
f_c = -\frac{P_e}{A_c} \pm \frac{P}{A_t}
$$
 (4-1.3)

Here,

 $A_t$  = transformed area of section

*P* = external axial force

*Pe* = effective prestress.

The external axial force is considered positive if it is tension and negative if it is compression. In the above expression, non-prestressed reinforcement is not considered. If there is non-prestressed reinforcement,  $A_c$  is to be substituted by  $(A_c +$  $(E_s/E_c)$   $A_s$ ) and  $A_t$  is to be calculated including  $A_s$ .

### **Analysis of Ultimate Strength**

The ultimate tensile strength of a section (*PuR*) is calculated as per **Clause 22.3**, **IS:1343 - 1980**. The ultimate strength should be greater than the demand due to factored loads.

In absence of non-prestressed reinforcement, the ultimate tensile strength of a section (*PuR*) is given as follows.

$$
P_{_{UR}} = 0.87 f_{_{pk}} A_{p}
$$
 (4-1.4)

In presence of non-prestressed reinforcement,

$$
P_{uR} = 0.87f_{y}A_{s} + 0.87f_{pk}A_{p}
$$
\n(4-1.5)

In the previous equations,

- $f_V$  = characteristic yield stress for non-prestressed reinforcement with mild steel bars
	- = characteristic 0.2% proof stress for non-prestressed reinforcement with high yield strength deformed bars.
- *fpk* = characteristic tensile strength of prestressing tendons.

The following example shows the design of a post-tensioned hanger for tension.

# **Example 4-1.1**

**Design a post-tensioned hanger to carry an axial tension of**  $P_{DL}$  **= 300 kN (dead** load including self-weight) and  $P_{LL}$  = 130 kN. The dimension of the hanger is 250  $\times$  250 mm<sup>2</sup>.

**Design the section without considering non-prestressed reinforcement. Tension is not allowed under service loads.** 

**The grade of concrete is M 35. The age at transfer is 28 days. Assume 15% long term losses in the prestress.** 

**The following properties of the prestressing strands are available from tests.** 



## **Solution**

Preliminary calculations at transfer

 $A_c \approx A$  = 250 × 250  $= 62,500$  mm<sup>2</sup>

Allowable stress for M35 concrete under direct compression at transfer

$$
f_{cc,all} = 0.8 \times 0.51 f_{ci}
$$
  
= 0.8 \times 0.51 \times 35  
= 14.3 N/mm<sup>2</sup>

Maximum prestressing force at transfer

$$
P_{0\,\text{max}} = f_c A_c
$$
  
= 14.3 × 62,500  
= 892,500 N

Preliminary calculations at service

$$
A_t \approx A = 250 \times 250
$$
  
= 62,500 mm<sup>2</sup>

Stress in concrete

$$
f_c = -\frac{P_e}{A_c} + \frac{P}{A_t}
$$

Allowable stress at service

Considering 15% loss

$$
f_{ct,all} = 0 \text{ N/mm}^2
$$

$$
P_e = 0.85 P_0
$$

Substituting the values

$$
0 = -\frac{0.85P_0}{A} + \frac{P}{A}
$$

Preliminary calculations at service (continued…)

Solving,

$$
0.85P_0 = P
$$
  

$$
P_0 = \frac{300 + 130}{0.85}
$$
  

$$
= 506 \text{ kN}
$$

Allowable prestress in tendon

$$
f_{p0} = 0.8f_{pk}
$$
  
= 0.8 × 1860  
= 1488 N/mm

= 591kN

 $2$ 

Required area of tendon

$$
A_p = \frac{506,000}{1488}
$$
  
= 340 mm<sup>2</sup>  
Select 4 strands with  

$$
A_p = 4 \times 99.3
$$
  
= 397.2 mm<sup>2</sup>  
Press the transfer  

$$
P_0 = 397.2 \times 1488 \text{ N}
$$

Indian Institute of Technology Madras

Final calculations at transfer

$$
A_c = 62,500 - 397
$$
  
= 62103 mm<sup>2</sup>

Stress in concrete

Stress in concrete

\n
$$
f_c = -\frac{P_0}{A_c}
$$
\n
$$
= -\frac{591,000}{62,103}
$$
\n
$$
= -9.5 \text{ N/mm}^2
$$
\nOK

\n
$$
|f_c| < f_{cc,all}
$$

Final calculations at service

$$
E_p = 195 \text{ kN/mm}^2
$$
  
\n
$$
E_c = 5,000\sqrt{35}
$$
  
\n
$$
= 29,580 \text{ N/mm}^2
$$
  
\n
$$
A_t = 62,103 + \frac{195}{29.6} \times 397.2
$$
  
\n
$$
= 64,720 \text{ mm}^2
$$

Stress in concrete

$$
f_c = -\frac{P_e}{A_c} + \frac{P}{A_t}
$$
  

$$
f_c = -\frac{0.85 \times 591,000}{62,103} + \frac{(300 + 130) \times 10^3}{64,720}
$$
  

$$
= -1.4 \text{ N/mm}^2
$$

No tensile stress in concrete. OK.

Final calculations for ultimate strength

$$
P_{uR} = 0.87 f_{pk} A_p
$$
  
= 0.87 × 1860 × 397.2 N  
= 643.0 kN

Demand under factored loads

$$
P_u = 1.5(300 + 130)
$$
  
= 645.0 kN  

$$
P_{uR} \approx P_u
$$
 OK

Designed cross-section



Nominal non-prestressed reinforcement is provided for resisting thermal and shrinkage cracks.

